



## NONLINEAR VIBRATIONS IN AUTOMOTIVE SUSPENSIONS

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**Abstract** *The object of the paper is to offer a brief survey of numerical simulations in nonlinear vibration systems. As nonlinear equations are difficult to solve, nonlinear systems are commonly approximated by linear equations. Recently, considerable attention has been paid towards approximate solutions for analytically solving nonlinear differential equation. A new analytical method has been presented for approximating the stationary response statistics of a class of DDOF non-linear dynamical systems under stochastic white-noise excitation. This paper considers the response of a nonlinearizear string to random excitation. The developme:nt of computational methods provided an opportunity to create various kinds of software which are able to analyse the properties of the above mentioned systems.*

**Keywords:** *Nonlinear vibration systems, numerical simulation.*

### 1. INTRODUCTION

Recently, considerable attention has been paid towards approximate solutions for analytically solving nonlinear differential equation. A new analytical method has been presented for approximating the stationary response statistics of a class of DDOF non-linear dynamical systems under stochastic white-noise excitation. The most commonly applied and convenient procedure uses Yingfang, L. Zhao, Q. Chen suggestion to estimate the linearization coefficients in context with Wen's introduction of an analytical expression for the restoring force. In contrast to existing methods based on minimizing the error in the stochastic equation of motion, the proposed method minimizes a measure of the In particular, it gives considerably better estimates for the mean-square response than the partial linearization method. While some particular exact solutions are available for specific systems under white noise excitations, many practical problems have been handled by approximate approaches such as linearization and closure assumptions.

### 2 SYSTEM MODEL

This method has seen the broadest application because of their ability to accurately capture the response statistics [1,3,4] over a wide range of response levels while maintaining relatively light computational burden. The method will be briefly discussed in the following sections To illustrate the procedure of equivalent linearization theory, let us consider the following oscillator with a nonlinear damping force component. The excitation of the system assumed to be a Gaussian white noise with a constant spectrum  $S_0$ . The simplified 4-degree-of-freedom model [3] of a suspension system is shown in Fig.1. Model the transverse motion of a vehicle due to a bump in the road as the response due to an impulse of magnitude  $I$  applied to the front. Equations of motion [3,5,6], while neglecting very small terms we get

$$\begin{pmatrix} v^2 M + I & uvM - I & 0 & 0 \\ (u+v)^2 & (u+v)^2 & 0 & 0 \\ uvM + I & v^2 M + I & 0 & 0 \\ (u+v)^2 & (u+v)^2 & 0 & 0 \\ 0 & 0 & m & 0 \\ 0 & 0 & 0 & m \end{pmatrix} \begin{pmatrix} \ddot{y}_1 \\ \ddot{y}_2 \\ \ddot{y}_3 \\ \ddot{y}_4 \end{pmatrix} + \begin{pmatrix} c_1 & 0 & -c_1 & 0 \\ c_2 & 0 & -c_2 & 0 \\ -c_3 & 0 & c_1 + c_3 & 0 \\ 0 & -c_2 & 0 & c_2 + c_4 \end{pmatrix} \begin{pmatrix} \dot{y}_1 \\ \dot{y}_2 \\ \dot{y}_3 \\ \dot{y}_4 \end{pmatrix} +$$

$$\begin{matrix} + \\ + \end{matrix} \begin{pmatrix} k_1 & 0 & -k_1 & 0 \\ 0 & k_2 & 0 & -k_2 \\ -k_1 & 0 & k_1+k_3 & 0 \\ 0 & -k_2 & 0 & k_2+k_4 \end{pmatrix} \begin{pmatrix} y_1 \\ y_2 \\ y_3 \\ y_4 \end{pmatrix} + r \begin{pmatrix} k_1 & 0 & -k_1 & 0 \\ 0 & k_2 & 0 & -k_2 \\ -k_1 & 0 & k_1+k_3 & 0 \\ 0 & -k_2 & 0 & k_2+k_4 \end{pmatrix} \begin{pmatrix} y_1 \\ y_2 \\ y_3 \\ y_4 \end{pmatrix} = \begin{pmatrix} \frac{W(t)}{4} \\ \frac{W(t)}{4} \\ \frac{W(t)}{4} \\ \frac{W(t)}{4} \end{pmatrix}. \quad (1)$$

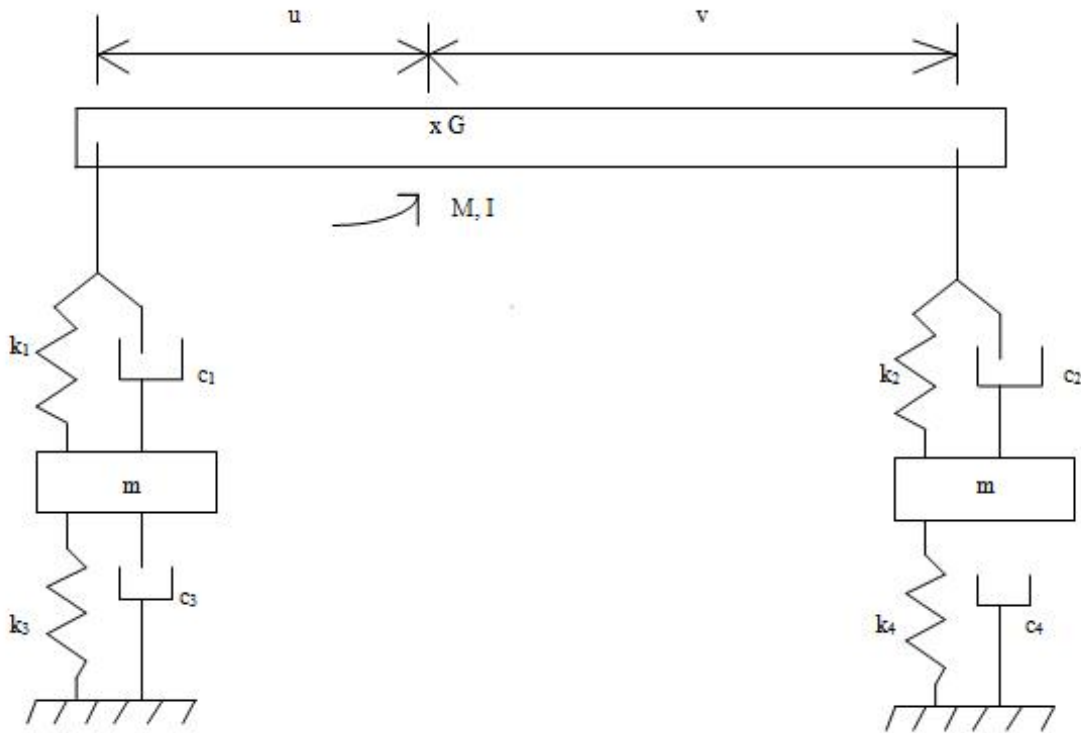
$W(t)$  is the external excitation signal with zero mean and  $y(t)$  is the displacement response of the system. The signal  $y(t)$  has a finite number of finite discontinuities [1,6,7] and contains a finite number of maxima and minima.

For large values of the  $\dot{y}$  damping force is positive, the curve has a positive slope and energy is removed from the system. On the other hand, for small values of the  $\dot{y}$  damping force is negative, the curve has a negative slope and energy is put into the system.

The stochastic linearization technique can be considered to be an extension of the equivalent linearization method for the treatment of nonlinear systems under deterministic excitations.

Based on the numerical studies of the rolling ship motion and the non-linearly damped Duffing oscillator [2,7,9,10], the proposed method is shown to be promising for providing computationally efficient and relatively accurate estimates of the stationary PDF, the mean-square response, and the out crossing rates.

The idea of linearization [13,14,15] is replacing the equation by a linear system. The linear equation can be write



**Figure. 1:** 4-degree-of-freedom model of a suspension.

$$\begin{pmatrix} \frac{v^2 M + I}{(u+v)^2} & \frac{uvM - I}{(u+v)^2} & 0 & 0 \\ \frac{uvM + I}{(u+v)^2} & \frac{v^2 M + I}{(u+v)^2} & 0 & 0 \\ 0 & 0 & m & 0 \\ 0 & 0 & 0 & m \end{pmatrix} \begin{pmatrix} \ddot{y}_1 \\ \ddot{y}_2 \\ \ddot{y}_3 \\ \ddot{y}_4 \end{pmatrix} + \begin{pmatrix} c_1 & 0 & -c_1 & 0 \\ c_2 & 0 & -c_2 & 0 \\ -c_3 & 0 & c_1 + c_3 & 0 \\ 0 & -c_2 & 0 & c_2 + c_4 \end{pmatrix} \begin{pmatrix} \dot{y}_1 \\ \dot{y}_2 \\ \dot{y}_3 \\ \dot{y}_4 \end{pmatrix} + \begin{pmatrix} k_{1e} & 0 & -k_{1e} & 0 \\ 0 & k_{2e} & 0 & -k_{2e} \\ -k_{1e} & 0 & k_{1e} + k_{3e} & 0 \\ 0 & -k_{2e} & 0 & k_{2e} + k_{4e} \end{pmatrix} \begin{pmatrix} y_1 \\ y_2 \\ y_3 \\ y_4 \end{pmatrix} = \begin{pmatrix} \frac{W(t)}{4} \\ \frac{W(t)}{4} \\ \frac{W(t)}{4} \\ \frac{W(t)}{4} \end{pmatrix}. \quad (2)$$

The difference between the nonlinear stiffness [7,8] and linear stiffness terms is

$$e = \begin{pmatrix} e_1 \\ e_2 \\ e_3 \\ e_4 \end{pmatrix}, \quad (3)$$

where

$$e_1 = k_1 y_1 - k_1 y_3 + \Gamma k_1 y_1^3 - \Gamma k_1 y_3^3 - k_{1e} y_1 + k_{1e} y_3 \quad (4)$$

$$e_2 = k_2 y_2 - k_2 y_4 + \Gamma k_2 y_2^3 - \Gamma k_2 y_4^3 - k_{2e} y_2 + k_{2e} y_4 \quad (5)$$

$$e_3 = -k_1 y_1 + (k_1 + k_3) y_3 - \Gamma k_1 y_1^3 + \Gamma (k_1 + k_3) y_3^3 + k_{1e} y_1 - (k_{1e} + k_{3e}) y_3 \quad (6)$$

$$e_4 = -k_2 y_2 + (k_2 + k_4) y_4 - \Gamma k_2 y_2^3 + \Gamma (k_2 + k_4) y_4^3 + k_{2e} y_2 - (k_{2e} + k_{4e}) y_4 \quad (7)$$

The value of  $k_{1e}$ ,  $k_{2e}$ ,  $k_{3e}$ ,  $k_{4e}$  can be obtained [8] by minimizing the expectation of the square error

$$\frac{dE\{e_1^2\}}{dk_{1e}} = 0, \quad (8)$$

$$\frac{dE\{e_2^2\}}{dk_{2e}} = 0, \quad (9)$$

$$\frac{dE\{e_3^2\}}{dk_{3e}} = 0, \quad (10)$$

$$\frac{dE\{e_4^2\}}{dk_{4e}} = 0. \quad (11)$$

Because

$$E\{y^6\} = \frac{(\dagger_y \sqrt{2})^6}{\sqrt{f}} \left[ -\frac{1}{2} e^{-u^2} u^5 \Big|_{-\infty}^{\infty} + \frac{5}{2} \int_{-\infty}^{\infty} u^4 e^{-u^2} du \right] = \quad (12)$$

$$= \frac{5(\dagger_y \sqrt{2})^6}{2\sqrt{f}} \int_{-\infty}^{\infty} u^4 e^{-u^2} du = 5 \dagger^2 E\{y^4\}$$

$$E\{y^4\} = 3 \dagger^2 E\{y^2\}, \quad (13)$$

$$E\{y^2\} = \dagger^2, \quad (14)$$

obtain

$$k_{1e} = k_1 + \frac{3\Gamma k_1 (\dagger_1^4 + \dagger_3^4)}{\dagger_1^2 + \dagger_3^2} \quad (15)$$

$$k_{2e} = k_2 + \frac{3\gamma k_2 (\dagger_2^4 + \dagger_4^4)}{\dagger_2^2 + \dagger_4^2} \quad (16)$$

$$k_{3e} = k_3 + 3\gamma \dagger_3^2 (k_1 + k_3) \quad (17)$$

$$k_{4e} = k_4 + 3\gamma \dagger_4^2 (k_2 + k_4) \quad (18)$$

Using the Fourier transform [9,10] of equation (2.) and having the relations

$$F(\dot{y}_k(t)) = i\check{S} \bar{y}_k(\check{S}), \quad k=1, \dots, 4. \quad (19)$$

$$F(\ddot{y}_k(t)) = i\check{S} F(\dot{y}_k(t)) = -\check{S}^2 \bar{y}_k(\check{S}) \quad (20)$$

$$F(F(t)) = \bar{F}(\check{S}), \quad (21)$$

obtain the system

$$\left\{ \begin{array}{l} \bar{y}_1 \left( ic_1 \check{S} + k_{1e} - \frac{v^2 M + I}{(u+v)^2} \check{S}^2 \right) - \frac{uvM - I}{(u+v)^2} \check{S}^2 \bar{y}_2 - k_{1e} \bar{y}_3 = \frac{\bar{F}(\check{S})}{4} \\ \bar{y}_1 \left( ic_2 \check{S} - \frac{uvM + I}{(u+v)^2} \check{S}^2 \right) - \left( \frac{v^2 M + I}{(u+v)^2} \check{S}^2 + k_{2e} \right) \bar{y}_2 - ic_3 \check{S} \bar{y}_3 - k_{1e} \bar{y}_3 = \frac{\bar{F}(\check{S})}{4} \\ \bar{y}_1 (-ic_3 \check{S} + k_{1e}) + [-m\check{S}^2 + (c_1 + c_3)i\check{S} + (k_{1e} + k_{3e})] \bar{y}_3 = \frac{\bar{F}(\check{S})}{4} \\ \bar{y}_2 (-ic_2 \check{S} - k_{2e}) + [-m\check{S}^2 + (c_2 + c_4)i\check{S} + (k_{2e} + k_{4e})] \bar{y}_4 = \frac{\bar{F}(\check{S})}{4} \end{array} \right. \quad (22)$$

We obtain the frequency response function [1,2] of the system.

The mean square value for the displacement [3,9,10] of the system is given by equation

$$\dagger_{y_k}^2 = R_{y_k}(0) = \int_{-\infty}^{\infty} \left| H_k(\check{S}) \right|^2 m_k^2 S'_0 d\check{S} = \frac{1}{m_k^2} \int_{-\infty}^{\infty} \left| H_k(\check{S}) \right|^2 m_k^2 S'_0 d\check{S}. \quad (23)$$

The power spectral density of response [13,14] for the first structure ( in  $m^2 \cdot s$  ) is given by equation

$$S_k(\check{S}) = \left| H_k(\check{S}) \right|^2 S_F = \left| H_k(\check{S}) \right|^2 m_k^2 S'_0(\check{S}) = \left| \frac{1}{m_k} H_k(\check{S}) \right|^2 m_k^2 S'_0(\check{S}) = \left| H_1(\check{S}) \right|^2 S'_0(\check{S}) \quad (24)$$

### 3 NUMERICAL RESULTS

For example, we consider

$$u=2m, \quad v=2m, \quad m=25kg, \quad M=175kg, \quad k_1=k_2=3 \times 10^5 \frac{N}{m}, \quad k_3=k_4=1 \times 10^5 \frac{N}{m}, \quad (25)$$

$$I=150kg \times m^2, \quad c_1=c_2=3000 \frac{N \times s}{m}, \quad c_3=c_4=750 \frac{N \times s}{m}, \quad \gamma=97,7m^{-2}. \quad (26)$$

Substituting given values, the mass, damping, and stiffness matrices are

$$\mathbf{M} = \begin{pmatrix} 53,125 & 34,375 & 0 & 0 \\ 53,125 & 53,125 & 0 & 0 \\ 0 & 0 & 25 & 0 \\ 0 & 0 & 0 & 25 \end{pmatrix} \quad (27)$$

$$\mathbf{C} = \begin{pmatrix} 3000 & 0 & -3000 & 0 \\ 3000 & 0 & -3000 & 0 \\ -3000 & 0 & 3750 & 0 \\ 0 & -3000 & 0 & 3750 \end{pmatrix} \quad (28)$$

$$\mathbf{K} = \begin{pmatrix} 3 \times 10^5 & 0 & -3 \times 10^5 & 0 \\ 0 & 3 \times 10^5 & 0 & -3 \times 10^5 \\ -3 \times 10^5 & 0 & 4 \times 10^5 & 0 \\ 0 & -k_2 & 0 & 4 \times 10^5 \end{pmatrix} \quad (29)$$

The displacement variances [4,6] of the system under Gaussian white noise excitation can be expressed as,

$$\dagger_{y_1}^2 = 62,5 \cdot 10^{-5} m^2, \quad \dagger_{y_2}^2 = 62,5 \cdot 10^{-5} m^2, \quad (30)$$

$$\dagger_{y_1}^2 = 47 \cdot 10^{-5} m^2, \quad \dagger_{y_2}^2 = 47 \cdot 10^{-5} m^2, \quad (31)$$

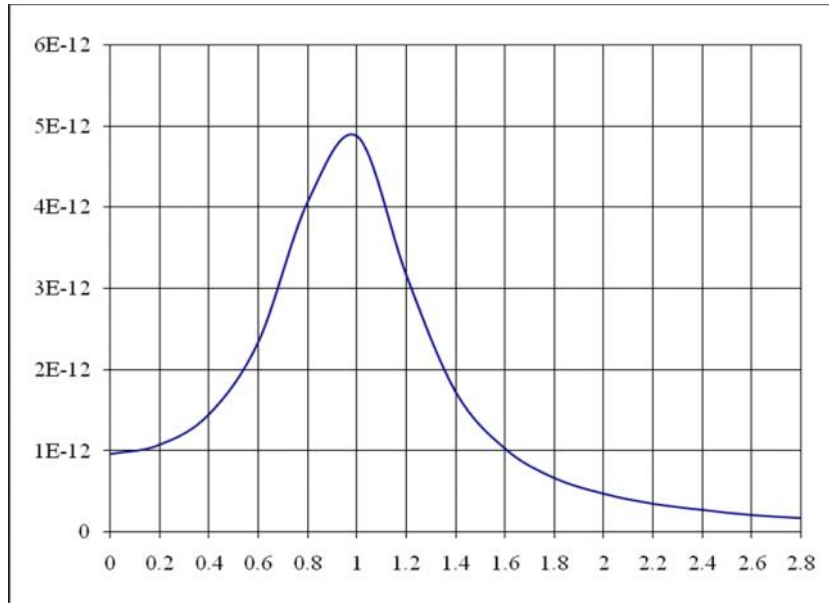
For the value of  $k_e$ , obtain

$$k_{1e} = k_1 + \frac{3r k_1 (\dagger_1^4 + \dagger_3^4)}{\dagger_1^2 + \dagger_3^2} = 3,48 \times 10^5 \frac{N}{m}, \quad (32)$$

$$k_{2e} = k_2 + \frac{3r k_2 (\dagger_2^4 + \dagger_4^4)}{\dagger_2^2 + \dagger_4^2} = 3,48 \times 10^5 \frac{N}{m}, \quad (33)$$

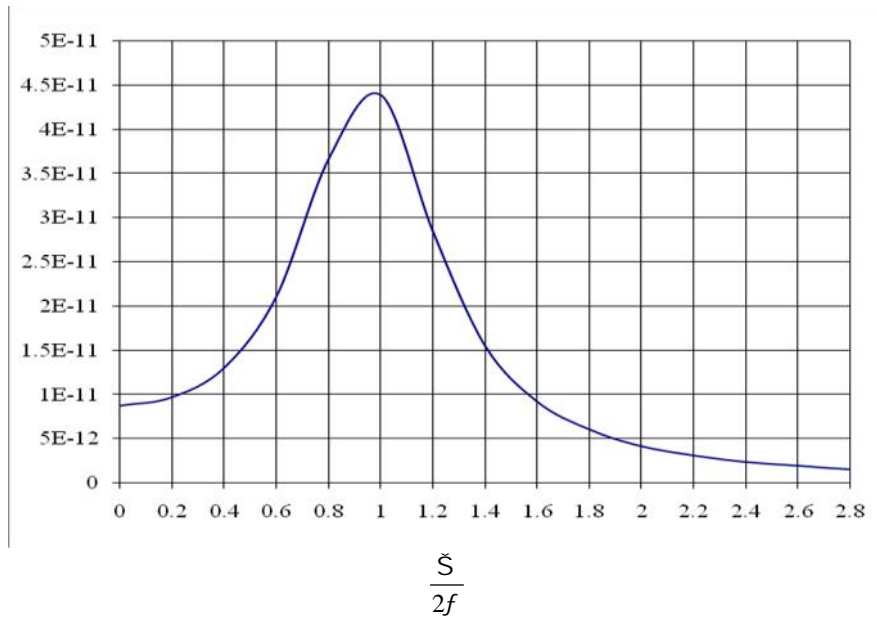
$$k_{3e} = k_3 + 3r \dagger_3^2 (k_1 + k_3) = 1,23 \times 10^5 \frac{N}{m}, \quad (34)$$

$$k_{4e} = k_4 + 3r \dagger_4^2 (k_2 + k_4) = 1,23 \times 10^5 \frac{N}{m}. \quad (35)$$



$$\frac{\xi}{2f}$$

**Figure 2:** The power spectral density of the response  $S_k(\xi) [m^2 \cdot s], k = 1, k = 2$ .



**Figure 3:** The power spectral density of the response  $S_k(\xi) [m^2 \cdot s], k = 3, k = 4$ .

#### 4. CONCLUSION

This nonlinearity is one of the reasons why accurate long-term forecasts are impossible with current technology. In research of vehicle system dynamics, investigation of parametral sensitivity of nonlinear systems is considered a very important question either in stability problems or in respect of chaotic behaviour. In studies of that purpose the computer aided numerical simulations are getting more and more important even if they consider relatively simple models.

#### REFERENCES

- [1] Pandrea, N., Parlac, S., Mechanical vibrations, Pitesti University, 2000.
- [2] A. Blaquiere, Nonlinear System Analysis, Academic Press, New York, 1966.
- [3] Graham, S., Theory and problems of mechanical vibrations, 1993
- [4] N. V. Butenin, Elements of the Theory of Nonlinear Oscillations, Blaisdell Publishing, New York, 1965
- [5] Stan, P., Stan, M., The power spectral density of response for non-linear structure to random vibration stages, 4<sup>th</sup> International Conference, Computational Mechanics And Virtual Engineering, Brasov, 2011.
- [6] Stan, P., Stan, M., On response of random vibration for nonlinear systems, 5<sup>th</sup> international conference Advanced Composite Materials Engineering, Brasov, 2014.
- [7] Stan, P., Stan, M., The Random vibrations for the nonlinear oscillators with an elastic sine-like feature, 5<sup>th</sup> international conference, Thermal Systems and Environmental Engineering, Galati, 2014.
- [8] Stan, P., "Random vibrations of non-linear oscillators", Second International Conference of Romanian Society of Acoustics on Sound and Vibration, october 2004
- [9] Stan, P., Stan, M., Random vibration of single-degree of freedom systems with stationary response, International Conference of mechanical engineering, ICOME, Craiova, 2013.
- [10] Stan, P., "Response of Duffing Oscillator under narrow band random excitation", The book of University by Pite ti, Serie Appl. Mech. 6, 2007.
- [11] Stan, P., "Analysis of single-degree of freedom non-linear structure under Gaussian white noise ground excitation", The book of University by Pite ti, Serie Appl. Mech. 6, 2007
- [12] Clough, R., W., Penzien, J., Dynamics of Structures, McGraw-Hill, New York, 1993.
- [13] Zhu, W., Q., Stochastic averaging method in random vibration, Bulletin S.F.M, 5, 1988.
- [14] Roberts, J.B., and Spanos, P.D., Random Vibration and Statistical Linearization, John Wiley, New York, 1990.
- [15] Roberts, J., B., First passage probability for nonlinear oscillators'' J. Engng Mech, ASCE, 102, 1976.